

Taking Generative Design to the Next Level on a LARGE Jet Engine Component

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#### **Agenda**

- Project motivation
- Eco-design engine demonstrator
- Multi-physics Generative Design
- Design for Additive Manufacturing
- Summary

## Project 'Monaco': Manufacturing of a Large-**Scale AM Component**

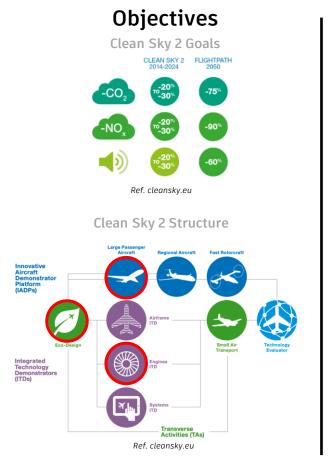


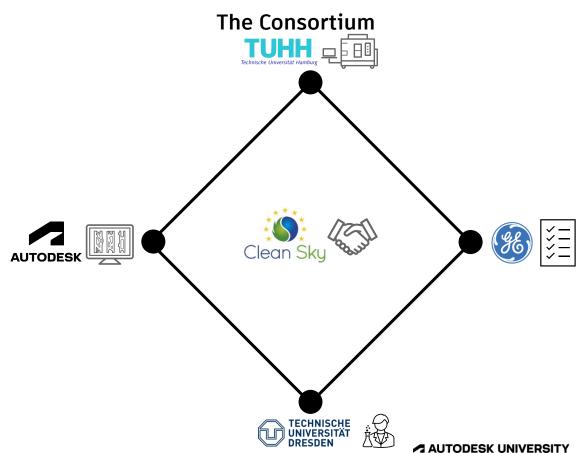


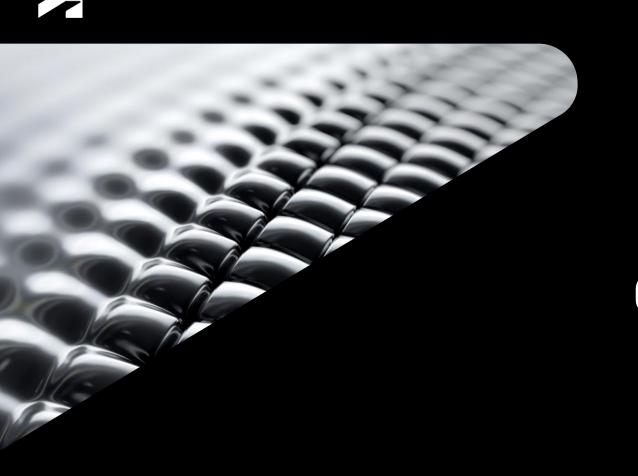


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#### Where Innovation Takes Off



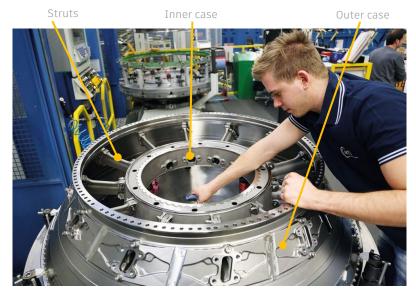




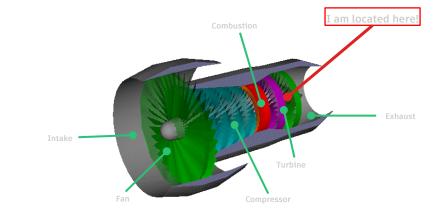
# Goals and Challenges

#### What Am I?

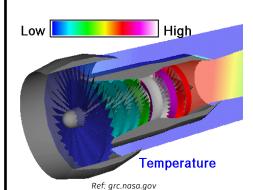
#### **Turbine Center Frame Assembly**



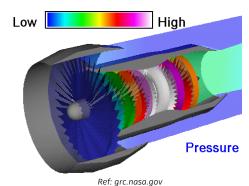
The turbine center frame of a GEnx engine is a demonstration component of the project partner MTU Aero Engines for hybrid additive manufacturing by laser material deposition. Photo via MTU Aero Engines.



Ref: grc.nasa.gov



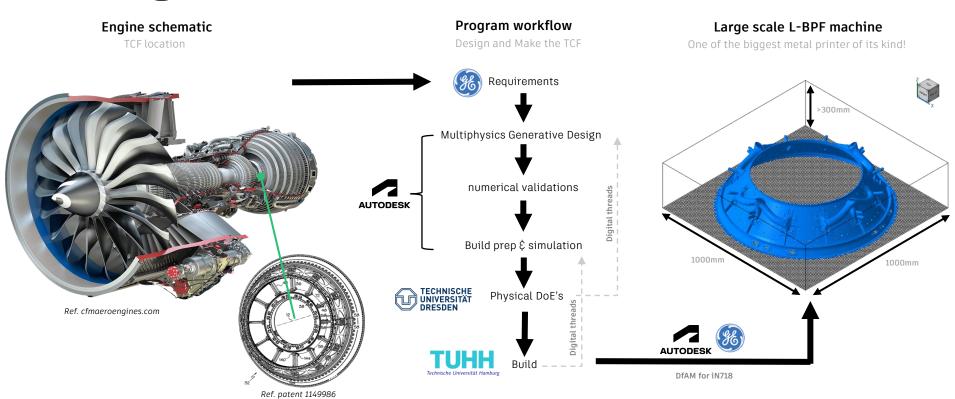
Max operating Temperature = <400°C



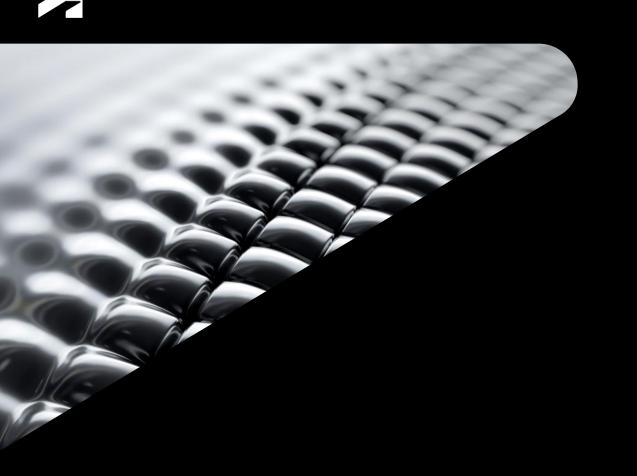
Max operating radial loading = >600kN

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### Making a Leaner, Cleaner and Greener Aircraft



Objectives: Reduce mass, increase performance, add functionality, reduce cost and for large scale AM



# Generative Design

#### **Design Evolution**

Design 1

Skin Optimization Philosophy



- 100+ to 1 part consolidation
- Mass reduction: 28.9%
- Pressure drop: 58%

#### Design 2

Manual design Philosophy



- 100+ to 1 part consolidation
- Mass reduction: 29.4%
- Pressure drop: 69%

#### Design 3

GD Inspired 'ribs' Philosophy



- 100+ to 1 part consolidation
- Mass reduction: 21%
- Pressure drop: 71%
- Meets stiffness requirements

#### Design 4.1 & 4.2

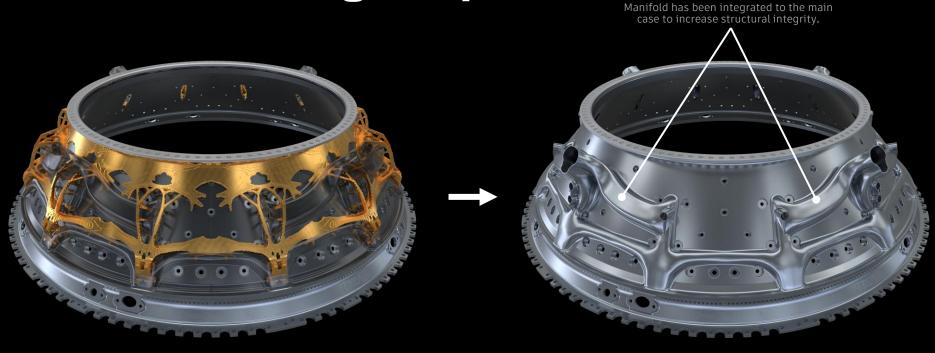
GD Inspired 'lattice' Philosophy

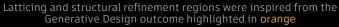




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# Generative Design Inspiration





100+ parts have been consolidated to one part!



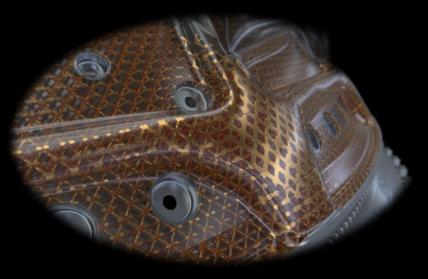




# **Lattice Generation**

Double Conformal Latticing was used to increase stiffness to weight ratio





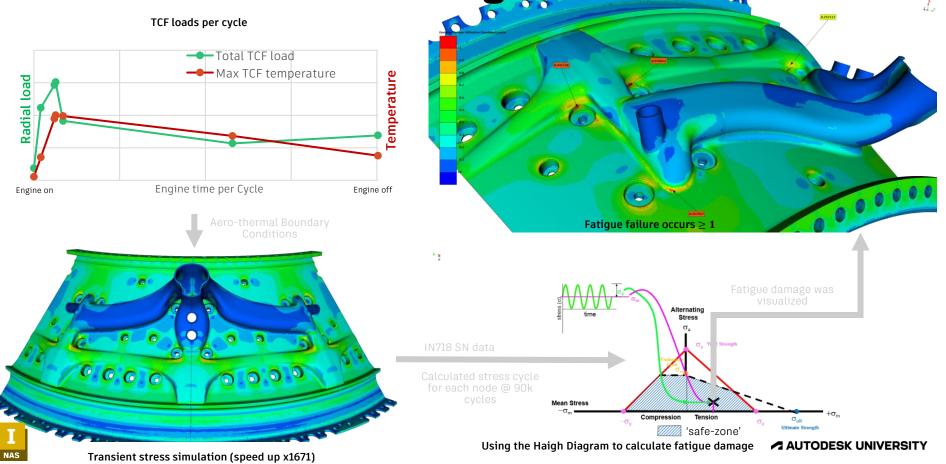
Variable beam density – structural reinforcement





- Mass reduction is 35%
- Meets stiffness (max displacement ≤ **37microns**)

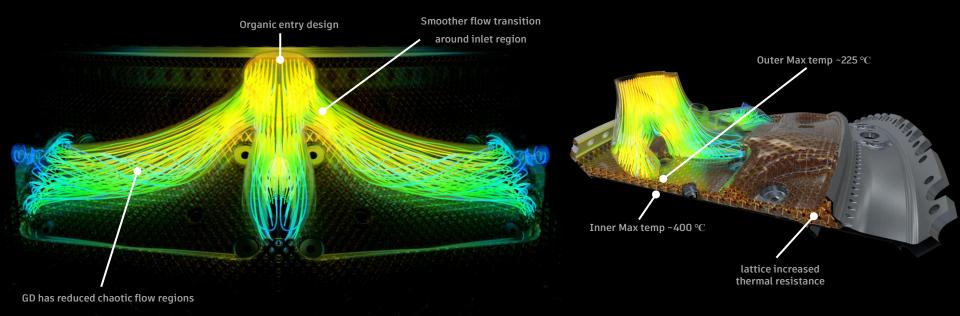
Thermal-Mechanical Fatigue Assessment



## Generative Fluids + Thermal Energy Savings

**Generative fluids** 

Lattice insulation



- Pressure drop has reduced by 91%
  - Flow outlets remained balanced

Lattice design has saved
 16GJ of heat energy







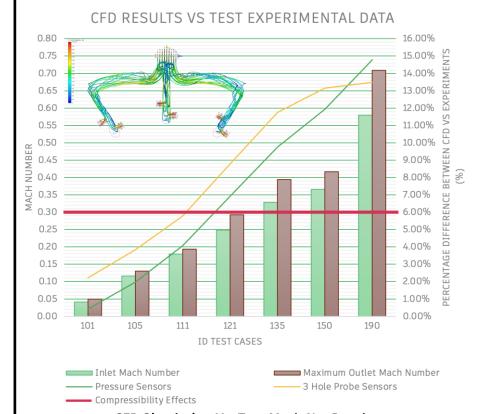
#### **Aerothermal Testing**



Manifold was manufactured out of two parts



Assembled and instrumented test rig





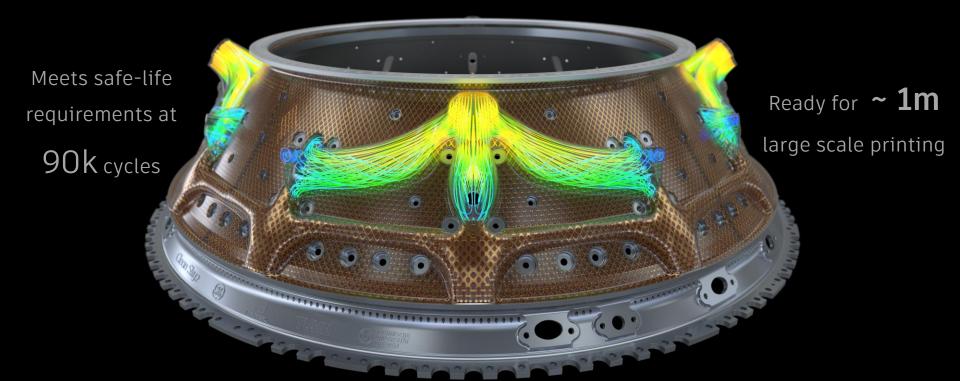




#### **Design 4.2 Summary**

**100+** parts consolidated into **1** 

~16GJ of heat has been saved





System pressure drop: ~91%

Overall mass reduction: ~35%

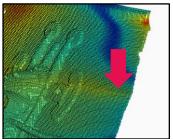




## **Manufacturing Design Evolution**

**Design 2**Manual Design Philosophy



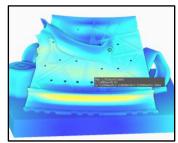


- Manufacture first 1/8 section of TCF
- First comparison between printed + scanned part and build process simulation

Design 3

GD Inspired 'Ribs' Philosophy



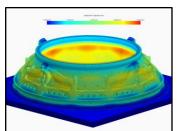


- Improved manufacturability with less distortion through the added stiffness rips
- Build process simulation and validation

Design 4.1

GD Inspired 'Lattice' Philosophy



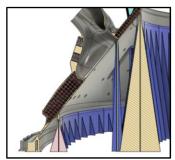


- Meets the stiffness and weight requirement
- Conformal lattice following the cone shape
- Full scale print simulation for large scale laser powder bed fusion process

Design 4.2

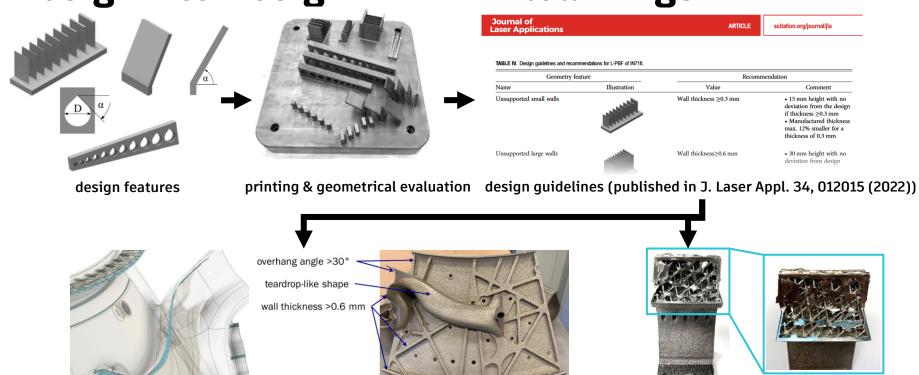
GD Inspired 'lattice' + DFAM Guidelines





 Applied DFAM guidelines for IN718 on large scale powderbed fusion machine

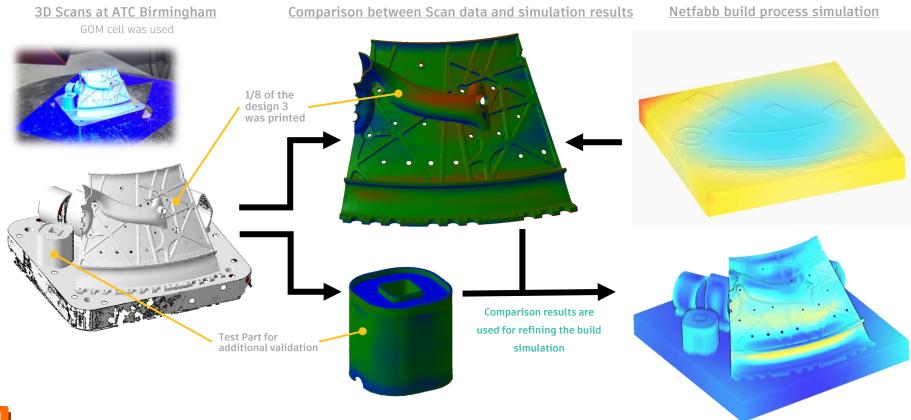
## Design 2 to Design 4 - DfAM learnings



application to TCF case & validation on design 3 and design 4.x

validation of lattice printing and powder removal process for design 4.x AUTODESK UNIVERSITY

## Design 3 - Build process simulation validation

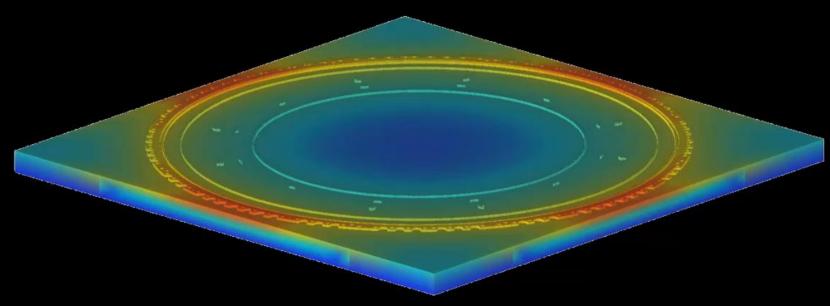


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# Design 4.1- Build process simulation

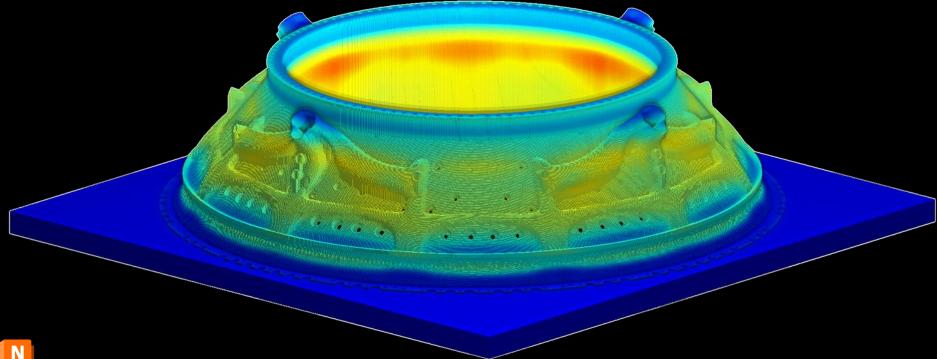
Full TCF simulation was done to identify potential problem areas





# Design 4.1- Build process simulation

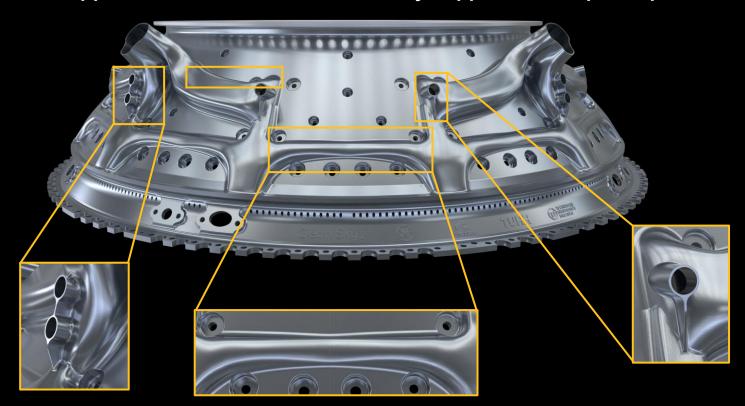
Full TCF simulation was done to identify potential problem areas





## Design 4.2 – Final DfAM applications

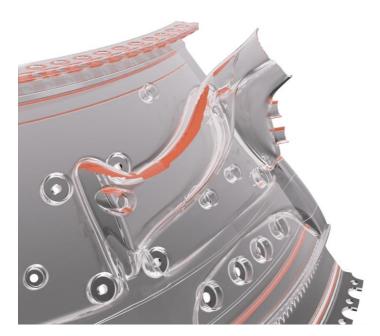
Further DFAM applied at R4.2 to reduce necessary support and improve problem areas



#### **Design 4.2 - Manifold modifications**

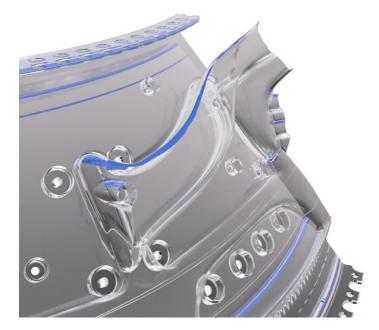
Reduced radii at top to avoid internal support requirements

Design 4.1



Total Predicted support structure volume: 965 cm<sup>3</sup>

Design 4.2





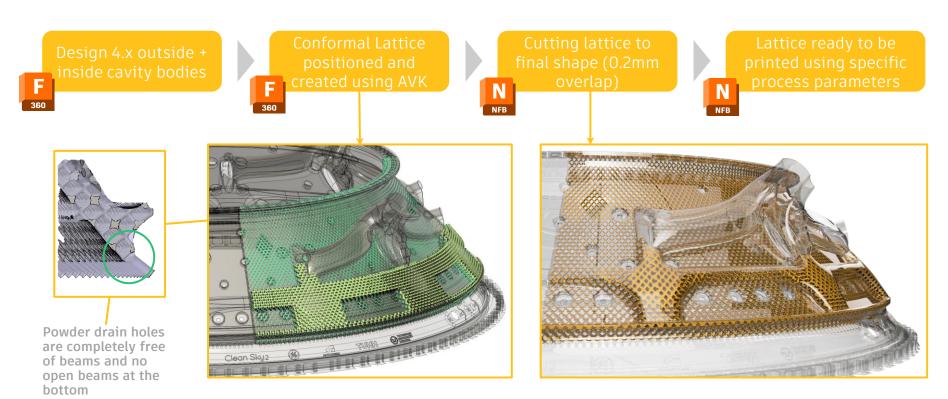






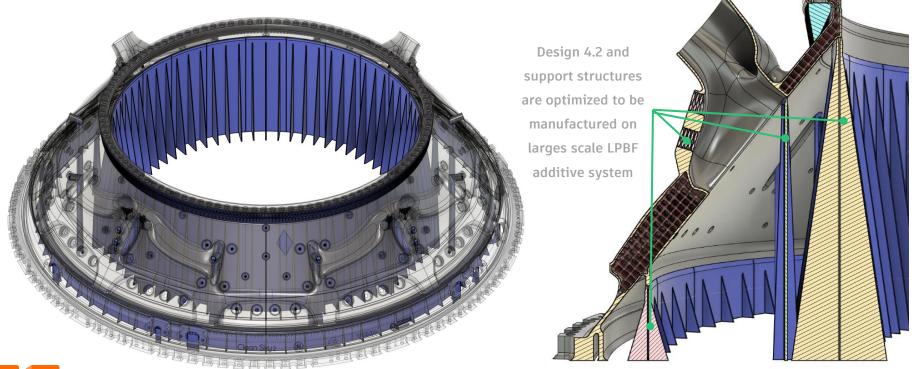
#### Design 4.x - Conformal lattice workflow

Conformal X-lattice around the whole hardware with variable beam thicknesses



#### Design 4.2 – final build preparation

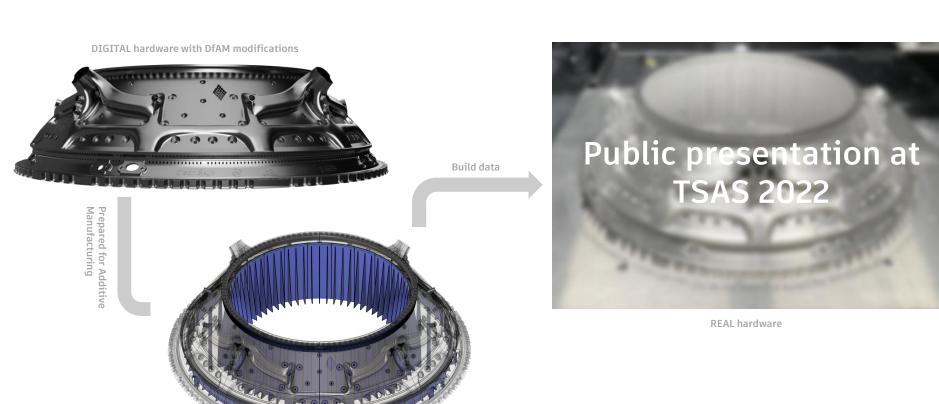
Break off support structures for large scale powder bed fusion process







## Printing Design 4.2 on Large scale AM system



# **Project MonAco Summary**



#### Part design

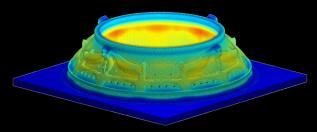
- Structural and fluid optimization of the turbine hardware
- Reduce mass by 35% and meet aero-thermal requirements
- Parts consolidation 100+ to 1





#### **AM Process**

- Simulation of build process & validation
- DMLM (PBF-LB/M) of up to 1m part diameter
- Created design guidelines for IN718

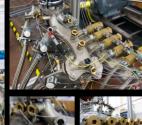




#### **Testing**

- Thermal and flow testing
- Mechanical properties
- Validation of Manufacturing
- TRL4/MRL4 at project end









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Thank you to everyone involved into the project over the last three years!!!



